

Integrate RCS, OMS, and MPS

Space Propulsion Technology  
Assessment Workshop

April 2001

## Current Baseline

### Integrate RCS, OMS, and MPS

- Separate RCS, OMS, and MPS Systems
- Different propellants for MPS versus RCS and OMS systems
- See “Integrate RCS and OMS” for RCS and OMS baseline
- MPS
  - High thrust oxygen/hydrogen or oxygen/HC main engines
  - One ox and one fuel tank
  - Autogenous pressurization of ox and fuel tanks
  - No connection to RCS or OMS systems

# Goal

## Integrate RCS, OMS, and MPS

- Desired gains
  - Only one ox and one fuel tank
    - Save as many as five ox and five fuel tanks
  - No separate pressurization systems
    - Save as many as five GHe and two GN2 tanks and seven pressurization feed systems
  - Two less major feed systems to develop
- Drawbacks
  - Some longer feed system lines
  - Some increased valve redundancy per feed line
    - Overall, fewer valves
  - If gas/gas OMS or RCS, then increased volume and weight of individual OMS or RCS feed lines
    - Overall volume and weight probably less than unintegrated system

## Potential Solutions

### Integrate RCS, OMS, and MPS

- Use same propellants for RCS and OMS as used by MPS
- Use one fuel and one ox accumulator to feed all RCS and OMS engines
  - Accumulator filled from MPS
  - Accumulator adds safety and functional redundancy
    - Size for deorbit with RCS
  - Can fill accumulator as part of start sequence
- Or accept lower specific impulse tank head pressure fed operation
  - May still be better specific impulse than current propellants
  - OMS engine will be large at such a low chamber pressure
  - May need screen or other propellant acquisition device for predictability


# Technology to Implement Solutions (TRLs)


## Integrate RCS, OMS, and MPS

- Ox/hydrogen or ox/HC OMS engines (4/5)
  - Gas/Gas
  - Gas/Liquid
  - Liquid/Liquid
- Ox/hydrogen or ox/HC RCS thrusters (4/5)
  - Gas/Gas
  - Gas/Liquid
  - Liquid/Liquid
- Blowdown accumulator OMS/RCS operations (4/5)
  - Filling from MPS
- Possibly propellant acquisition device (4/5)
- Combustion and ignition (4/5)
  - Impulse Bit
  - Ignition
  - Control system integration

Cost to Mature Technology

Integrate RCS, OMS, and MPS

\$100K	
\$500K	
\$1M	
\$5M	
\$10M	
\$30M	
\$50M	
\$100M	
\$500M	

6 Mo	
1 Yr	
18 Mo	
2 Yr	
3 Yr	
4 Yr	
5 Yr	
5 Yr+	